



The  
University  
Of  
Sheffield.

MAS380

SCHOOL OF MATHEMATICS AND STATISTICS

Autumn Semester 2017–18

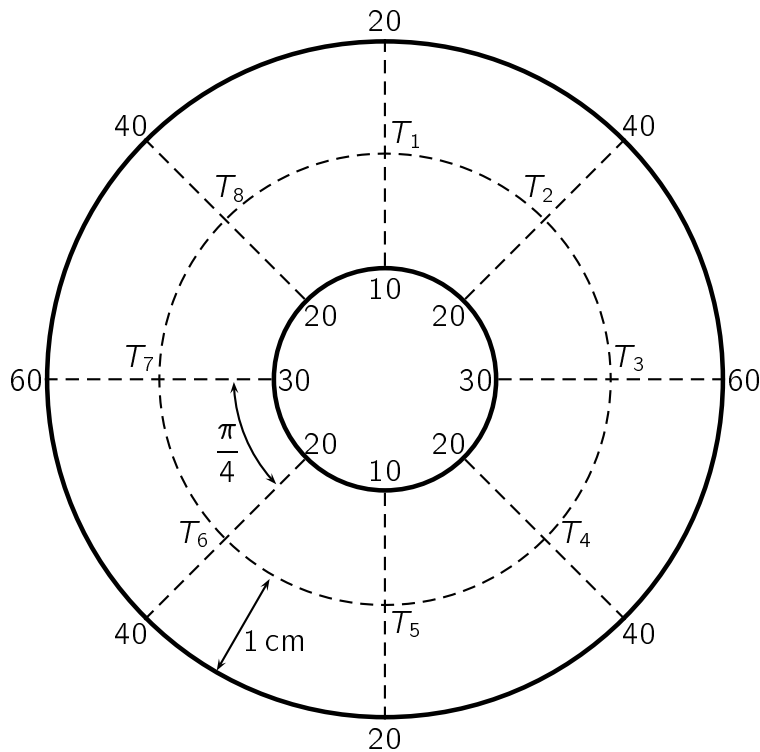
Computational Engineering Mathematics

Three hours

*Marks will be awarded for your best FOUR answers.  
The maximum possible mark for the paper is 100.*

- 1 The figure shows an annular plate (bounded by concentric circles of radius 1 cm and radius 3 cm) made of a homogeneous isotropic material. In polar coordinates  $(r, \theta)$  the plate is divided into intervals with spacing of  $\Delta r = 1$  cm in the radial ( $r$ ) direction, and  $\Delta\theta = \pi/4$  radians in the  $\theta$  direction. The temperature  $T(r, \theta)$  in this plate satisfies the indicated boundary conditions (given in  $^{\circ}\text{C}$ ) and has reached a steady-state condition so that it is described by Laplace's equation, which in polar coordinates is

$$\frac{\partial^2 T}{\partial r^2} + \frac{1}{r} \frac{\partial T}{\partial r} + \frac{1}{r^2} \frac{\partial^2 T}{\partial \theta^2} = 0.$$



- (a) Draw a sketch of the solution domain, showing clearly the lines of symmetry for  $T(r, \theta)$ , and indicating which of the unknown temperatures  $T_1, T_2, \dots, T_8$  are equal to each other. **(5 marks)**

1 (continued)

(b) If  $\Delta\theta$  is small and  $T_{i,j} = T(r_i, \theta_j)$ , give an argument for why

$$\frac{1}{r^2} \frac{\partial^2 T}{\partial \theta^2}(r_i, \theta_j) \approx \frac{1}{r_i^2} \frac{T_{i,j+1} - 2T_{i,j} + T_{i,j-1}}{(\Delta\theta)^2}. \quad (2 \text{ marks})$$

(c) Using the approximate result in part (b), and a central difference for  $\frac{\partial T}{\partial r}$ , find the finite difference equations required to find estimates of the nodal temperatures  $T_1$ ,  $T_2$  and  $T_3$ . Show that these can be expressed in the form  $A\mathbf{T} = \mathbf{b}$ , where

$$A = \begin{bmatrix} 16 + 4\pi^2 & -16 & 0 \\ -8 & 16 + 4\pi^2 & -8 \\ 0 & -16 & 16 + 4\pi^2 \end{bmatrix}, \quad \mathbf{T} = \begin{bmatrix} T_1 \\ T_2 \\ T_3 \end{bmatrix} \quad \text{and} \quad \mathbf{b} = \begin{bmatrix} 65\pi^2 \\ 130\pi^2 \\ 195\pi^2 \end{bmatrix},$$

with units in  $^\circ\text{C}$ .

(10 marks)

Hence, using Gaussian elimination or otherwise and taking  $\pi^2$  to be given by 10, find the values of  $T_1$ ,  $T_2$  and  $T_3$  which follow *exactly* from these equations.

(8 marks)

- 2 The temperature  $T(x, t)$  satisfies the heat conduction equation

$$\frac{\partial T}{\partial t} = 2 \frac{\partial^2 T}{\partial x^2} \quad (0 \leq x \leq 1, \quad t > 0). \quad (1)$$

- (a) If  $T_{i,j} = T(x_i, t_j)$ , with  $i = 0$  and  $i = N$  corresponding to  $x = 0$  and  $x = 1$ , respectively, and  $j = 0$  corresponding to  $t = 0$ , use backward differences for time derivatives and central differences for space derivatives to derive the implicit scheme

$$-2k T_{i+1,j} + (1 + 4k) T_{i,j} - 2k T_{i-1,j} = T_{i,j-1}$$

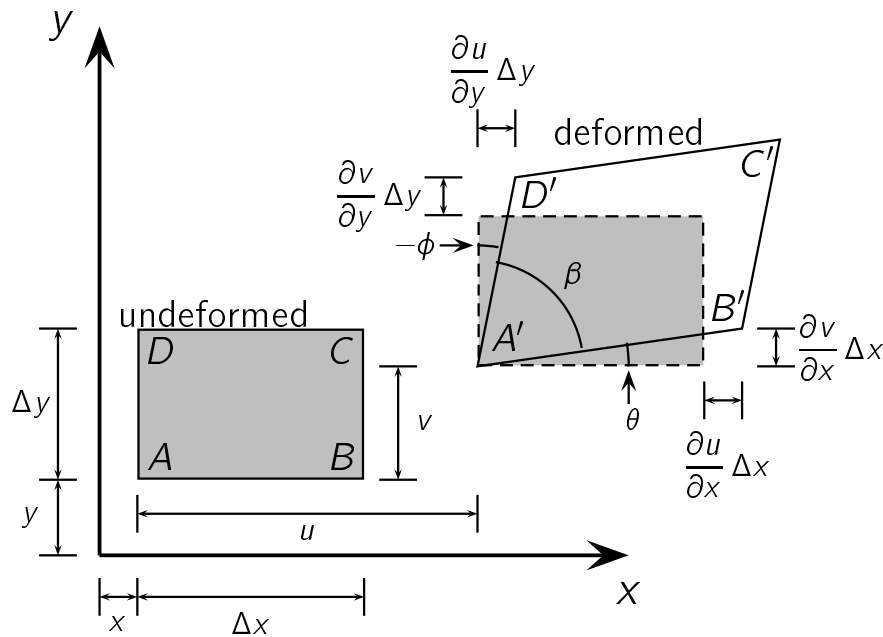
for  $i = 1, \dots, N - 1$  and  $j = 1, 2, \dots$ , where

$$k = \frac{\Delta t}{(\Delta x)^2}. \quad (4 \text{ marks})$$

- (b) Equation (1) is to be solved over  $0 \leq x \leq 1$ , with boundary conditions  $T(0, t) = 20$  and  $T(1, t) = 10$ , and initial temperature distribution  $T(x, 0) = 20 - 10x^2$ . Taking  $\Delta x = 0.2$  and  $\Delta t = 0.1$ , use the implicit scheme in part (a) to write down the system of equations for the temperature at  $x = 0.2, 0.4, 0.6, 0.8$  and time  $t = 0.1$ . (Note that you *do not* need to solve the equations.) **(11 marks)**
- (c) Show from your answer to part (b) that the Jacobi iteration equations to find the  $(n + 1)$ th iteration from the  $n$ th iteration are

$$\begin{bmatrix} T_{1,1} \\ T_{2,1} \\ T_{3,1} \\ T_{4,1} \end{bmatrix}^{(n+1)} = \frac{1}{11} \begin{bmatrix} 119.6 \\ 18.4 \\ 16.4 \\ 63.6 \end{bmatrix} + \frac{5}{11} \begin{bmatrix} 0 & 1 & 0 & 0 \\ 1 & 0 & 1 & 0 \\ 0 & 1 & 0 & 1 \\ 0 & 0 & 1 & 0 \end{bmatrix} \begin{bmatrix} T_{1,1} \\ T_{2,1} \\ T_{3,1} \\ T_{4,1} \end{bmatrix}^{(n)} \quad (10 \text{ marks})$$

- 3 (a) Referring to the figure, for small displacements  $u$  and  $v$ , define the normal strain  $\epsilon_{yy}$  and the engineering shear strain  $\gamma_{xy}$ . (3 marks)



Hence show that

$$\epsilon_{yy} = \frac{\partial v}{\partial y} \quad \text{and} \quad \gamma_{xy} = \frac{\partial v}{\partial x} + \frac{\partial u}{\partial y},$$

where  $u$  is the displacement of the body  $ABCD$  in the  $x$ -direction, and  $v$  is the displacement in the  $y$ -direction. (12 marks)

- (b) The matrix relating the *engineering* strains to the stresses for an isotropic material is given by

$$\begin{bmatrix} \lambda + 2\mu & \lambda & \lambda & 0 & 0 & 0 \\ \lambda & \lambda + 2\mu & \lambda & 0 & 0 & 0 \\ \lambda & \lambda & \lambda + 2\mu & 0 & 0 & 0 \\ 0 & 0 & 0 & \mu & 0 & 0 \\ 0 & 0 & 0 & 0 & \mu & 0 \\ 0 & 0 & 0 & 0 & 0 & \mu \end{bmatrix}$$

where

$$\lambda = \frac{E\nu}{(1+\nu)(1-2\nu)} \quad \text{and} \quad \mu = \frac{E}{2(1+\nu)}.$$

If  $E = 31.0$  GPa,  $\nu = 0.157$ ,  $\epsilon_{xx} = 927 \times 10^{-6}$ ,  $\epsilon_{yy} = 0.31\epsilon_{xx}$ ,  $\epsilon_{zz} = -0.23\epsilon_{xx}$ ,  $\epsilon_{xy} = 327 \times 10^{-6}$ ,  $\epsilon_{yz} = -453 \times 10^{-6}$  and  $\epsilon_{zx} = 53 \times 10^{-6}$  at a particular point in the material, calculate the stress at that point, giving your answers to 3 significant figures. (10 marks)

- 4 An incompressible fluid of constant density  $\rho$  is contained between vertical plane solid boundaries at  $x = -a$  and  $x = a$ , where  $a$  is a positive constant. The boundaries both move vertically upwards (in the  $z$ -direction) with constant speed  $U$ .

The pressure gradient in the vertical direction is zero, i.e.

$$\frac{\partial p}{\partial z} = 0,$$

and the fluid velocity  $\mathbf{v}$  is given by

$$\mathbf{v} = w(x) \mathbf{k},$$

where  $\mathbf{k}$  is the unit vector in the  $z$ -direction.

The velocity  $\mathbf{v}$  satisfies

$$\rho \frac{D\mathbf{v}}{Dt} = -\nabla p + \mu \nabla^2 \mathbf{v} - \rho g \mathbf{k},$$

where  $\mu$  is the constant viscosity,  $g$  is the constant gravitational acceleration, and

$$w = U \quad \text{when} \quad x = \pm a.$$

- (a) Show that

$$(\mathbf{v} \cdot \nabla) \mathbf{v} = \mathbf{0},$$

and hence that

$$\mathbf{0} = -\nabla p + \mu \frac{d^2 w}{dx^2} \mathbf{k} - \rho g \mathbf{k}. \quad (5 \text{ marks})$$

Find  $w(x)$ . (10 marks)

- (b) Find the fluid flux  $\int_{-a}^a w \, dx$  across a horizontal plane, per unit distance in the  $y$ -direction. (5 marks)

Show that if the flux is zero then

$$U = \frac{\rho g a^2}{3\mu}. \quad (2 \text{ marks})$$

Using this, find the fluid velocity midway between the boundaries.

(3 marks)

- 5 (a) A rectangular surface on a positive  $x$ -plane is defined by  $x = 0$ ,  $0 \leq y \leq 1$  and  $0 \leq z \leq 2$  (where all lengths are measured in m). The stress tensor depends on  $y$  and is given by (in units of Pa)

$$[\sigma] = \begin{bmatrix} 6y^2 & 2y & 4y^3 \\ 2y & 1 & -3y^2 \\ 4y^3 & -3y^2 & 1 \end{bmatrix}.$$

Show that the total stress force  $\mathbf{f}$  on the rectangular surface is

$$\mathbf{f} = 2 \begin{bmatrix} 2 \\ 1 \\ 1 \end{bmatrix} \text{ N.} \quad (6 \text{ marks})$$

- (b) In index notation the  $i$ -component of the curl of a vector field  $\mathbf{F}$  may be expressed as

$$(\nabla \times \mathbf{F})_i = \varepsilon_{ijk} \frac{\partial F_k}{\partial x_j},$$

where  $\varepsilon_{ijk}$  is the Levi-Civita tensor.

Show, using index notation, that

$$\nabla \times (\nabla \times \mathbf{F}) = \nabla(\nabla \cdot \mathbf{F}) - \nabla^2 \mathbf{F}. \quad (10 \text{ marks})$$

(You may use  $\varepsilon_{kij}\varepsilon_{klm} = \delta_{il}\delta_{jm} - \delta_{im}\delta_{jl}$ , where  $\delta_{ij}$  is the Kronecker delta tensor.)

- (c) (i) Using index notation, show that for any vector fields  $\mathbf{a}$ ,  $\mathbf{b}$  and  $\mathbf{c}$

$$(\mathbf{a} \times \mathbf{b}) \cdot \mathbf{c} = (\mathbf{b} \times \mathbf{c}) \cdot \mathbf{a}$$

and

$$\nabla \cdot (\mathbf{a} \times \mathbf{b}) = \mathbf{b} \cdot (\nabla \times \mathbf{a}) - \mathbf{a} \cdot (\nabla \times \mathbf{b}). \quad (6 \text{ marks})$$

- (ii) If  $V$  is a closed volume bounded by a surface  $S$ , Gauss' theorem states that

$$\int_V \nabla \cdot \mathbf{F} \, dV = \int_S \mathbf{F} \cdot \hat{\mathbf{n}} \, dS$$

for a vector field  $\mathbf{F}$ , where  $\hat{\mathbf{n}}$  is the unit normal to  $S$ , pointing out of  $V$ .

By choosing

$$\mathbf{F} = \mathbf{a} \times \mathbf{u},$$

where  $\mathbf{a}$  is an arbitrary constant vector, and using the results of part (i), show that

5 (continued)

$$\int_V \nabla \times \mathbf{u} \, dV = \int_S \hat{\mathbf{n}} \times \mathbf{u} \, dS. \quad (3 \text{ marks})$$

End of Question Paper



# Formula Sheet

Notation:

$$U(x_i, t_j) \equiv U_{i,j}$$

Forward difference formula for  $\partial U/\partial t$ :

$$\frac{\partial U}{\partial t}(x_i, t_j) \approx \frac{U_{i,j+1} - U_{i,j}}{\Delta t}$$

Forward difference formula for  $\partial U/\partial x$ :

$$\frac{\partial U}{\partial x}(x_i, t_j) \approx \frac{U_{i+1,j} - U_{i,j}}{\Delta x}$$

Backward difference formula for  $\partial U/\partial t$ :

$$\frac{\partial U}{\partial t}(x_i, t_j) \approx \frac{U_{i,j} - U_{i,j-1}}{\Delta t}$$

Backward difference formula for  $\partial U/\partial x$ :

$$\frac{\partial U}{\partial x}(x_i, t_j) \approx \frac{U_{i,j} - U_{i-1,j}}{\Delta x}$$

Central difference formula for  $\partial U/\partial x$ :

$$\frac{\partial U}{\partial x}(x_i, t_j) \approx \frac{U_{i+1,j} - U_{i-1,j}}{2\Delta x}$$

Central difference formula for  $\partial^2 U/\partial x^2$ :

$$\frac{\partial^2 U}{\partial x^2}(x_i, t_j) \approx \frac{U_{i+1,j} - 2U_{i,j} + U_{i-1,j}}{(\Delta x)^2}$$

**Relation between different parameters:**

A number of relationships between  $E$ ,  $\nu$ ,  $K$ ,  $\lambda$  and  $\mu$  hold and are summarized in Table 1.  $\mu$  ( $\equiv G$ ) is the elastic shear modulus,  $K$  the elastic bulk modulus,  $E$  the elastic stiffness (or Young's Modulus) and  $\nu$  Poisson's ratio.

	$E$	$\nu$	$K$	$\lambda$	$\mu \equiv G$
$E, \nu$	-	-	$\frac{E}{3(1-2\nu)}$	$\frac{E\nu}{(1+\nu)(1-2\nu)}$	$\frac{E}{2(1+\nu)}$
$E, K$	-	$\frac{3K-E}{6K}$	-	$\frac{K(9K-3E)}{9K-E}$	$\frac{3KE}{9K-E}$
$K, \mu$	$\frac{9\mu K}{3K+\mu}$	$\frac{3K-2\mu}{2(3K+\mu)}$	-	$K - \frac{2\mu}{3}$	-

Table 1: The relations between the properties of elastic bodies.